

Team II



**Team I
A
New Design Paradigm**

by
Dr. Knut I. Oxnevad

Jet Propulsion Laboratory
California Institute of Technology
Pasadena, CA

Pasadena, CA, January 20, 2000

1. In a Nutshell
2. Areas of Expertise
3. Approach
4. Conclusions & Summary

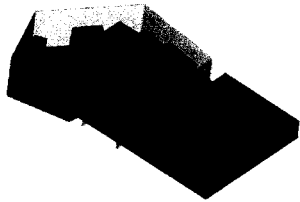
The work described in this presentation was carried out at the Jet Propulsion Laboratory, California Institute of Technology, under a contract with the National Aeronautics and Space Administration.

Team II

Approach



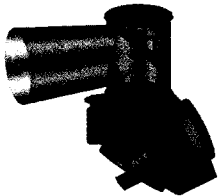
Discovery Phase 1
Gulliver



DS (ST)-4/CIRCLE



Search Camera for the
CNES Orbiter

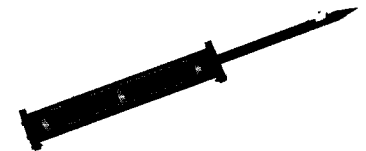


- *Concurrent Design Environment*
- *Total Systems Approach, Multi-Disciplinary Team*
- *Standing Design Team*
- *From Concept to Engineering Drawings*
- *Interconnected, High-End Optical, Microwave, Mechanical/CAD Thermal, Structural, Dynamics, Simulation, Orbital, Electronics Analysis and Design Tools, such as Code V, ZeMax, Mechanical Desktop, NASTRAN, Thermal Desktop, Adams, MODTool, and Working Model*
- *Applications Utilize a Common CAD Developed Geometry*
- *Open Environment, import/export of STEP, NASTRAN files, etc., from/to JPL, other NASA centers, and Industry*
- *Technology Insertion Through Cooperation with MDL/TAP*
- *Analysis and Design Time Cut from Months to Weeks*

IIP/OSIRIS

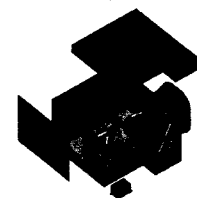


Loihi Deep Ocean,
Volcanic
Vent Probe

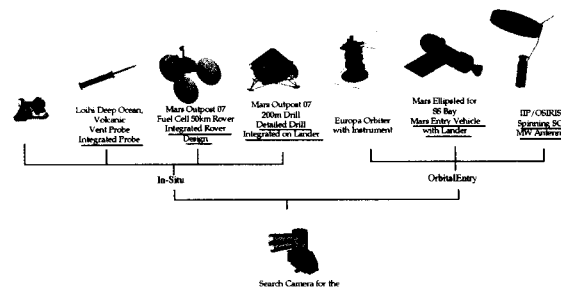


Team II

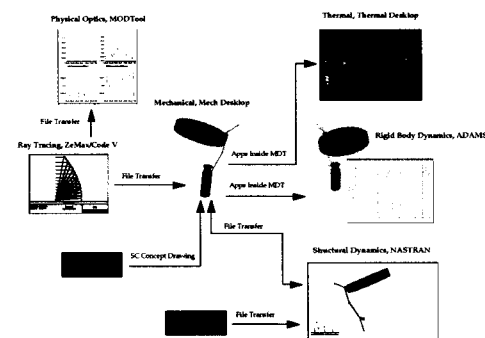
The Two Elements Function and Approach



1. Expertise

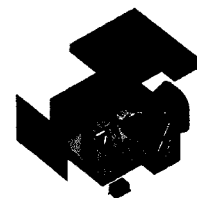


2. Approach (Design Paradigm)



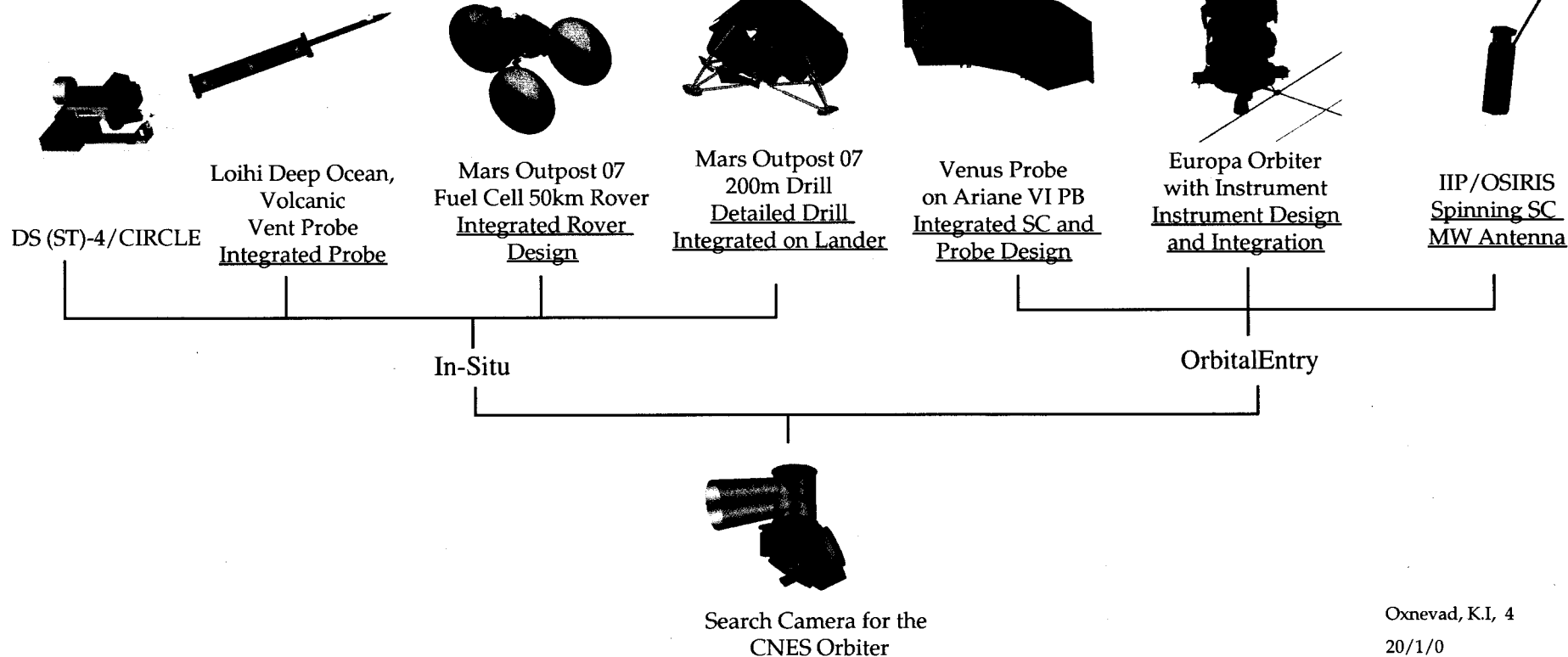
Team II

Support; Expertise



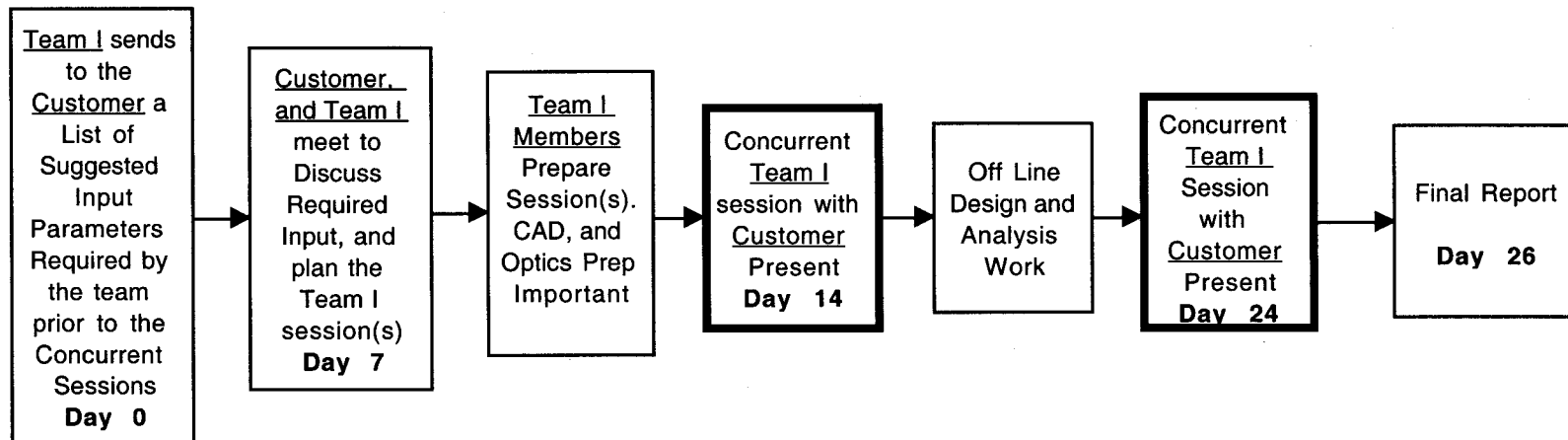
• Expertise

- Synthesis, Analysis, Simulation, and Design Support
- Orbital and In-situ Payloads
- Instruments to Fully Integrated Probes/Spacecraft
 - Optical, Microwave, Mass Spectrometer Instruments
 - Surface/Subsurface Probes, Rovers, Atmospheric Entry Vehicles, Dedicated SC.



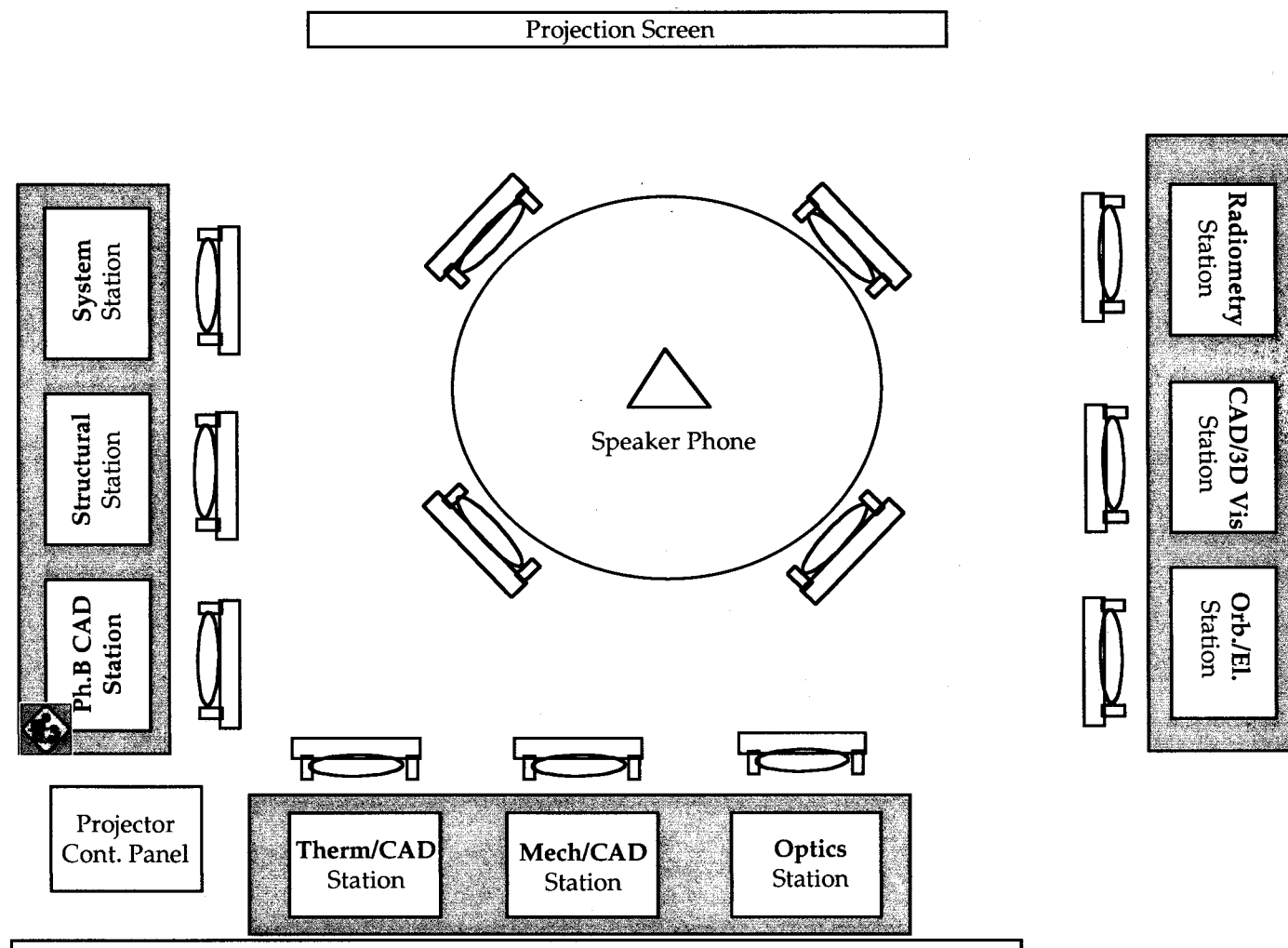
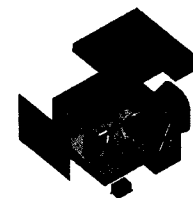
Team I

Overall Process



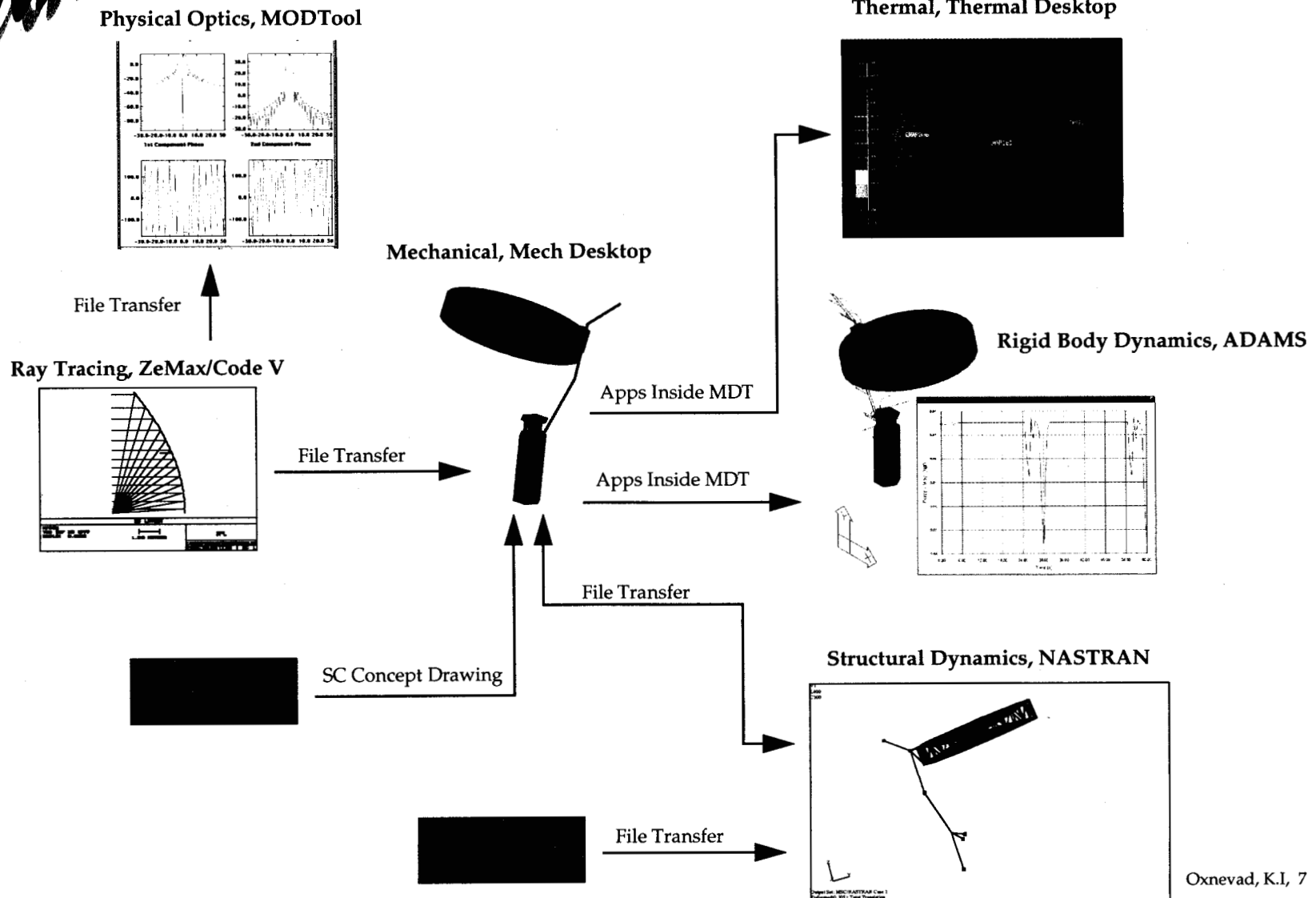
Team II

Stations



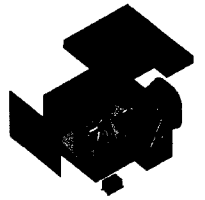
Team II

Approach (Design Paradigm): Integrated, High-End Analysis and Design



Team II

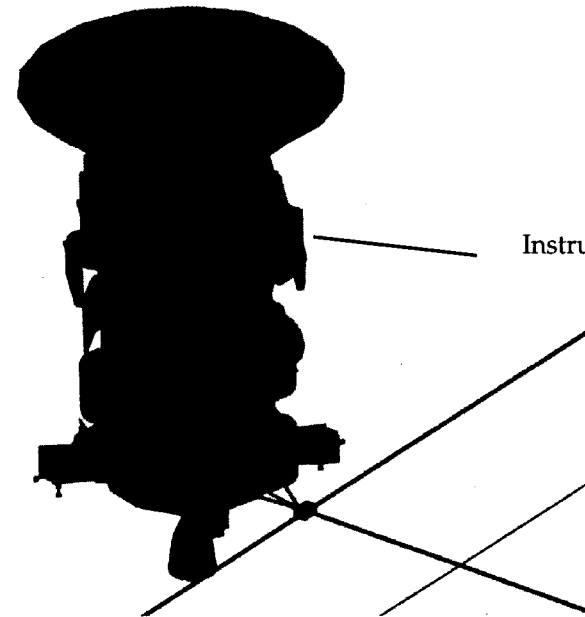
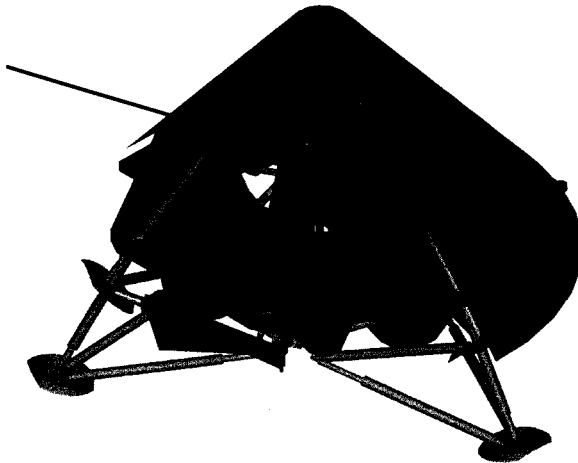
Approach Integration of Payload and SC/ Lander



Modified 03 Lander [STEP file]

Europa Orbiter [STEP file]

200m Drill
System



Instrument Box

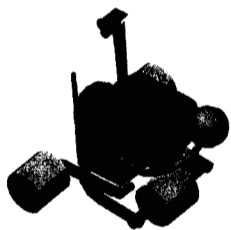
Team II

Approach Sizing and Configuration



**Mars Outpost
50km Fuel Cell Rover**

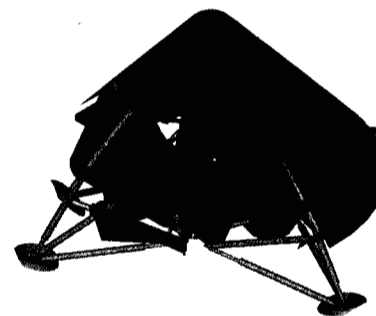
**Mars Outpost
200m Drill**



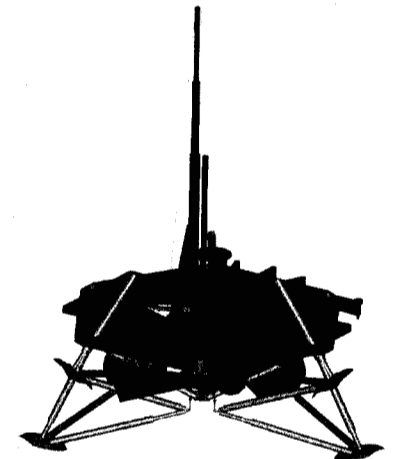
Lander Configuration



Surface Configuration



Packaged Configuration



Deployed Configuration



Summary and Conclusions



- *A New and Unique Design Paradigm*
- *Cusotmers Clearly See Benefits: Development Time Reduced and Quality Increased*
- *The Team I Environment Consequently Can Be Seen As a Laboratory for Developing Effective Conceptual Design Environments/Processes for Demanding Types of Space Instruments , Probes, Rovers , Other Types of Surface Systems, Telecomm Systems. and SC.*
- *Team I Related Procedures and Processes are Beginning to Radically Change the Instrument/Probe Design Process at JPL.*
- *The Concurrent Design Paradigm and Design Approaches Discussed here have the Potential of Bringing Great Benefits to any Large and Complex Design and Analysis Problem*